



IN APPLIED SCIENCE & ENGINEERING TECHNOLOGY

Volume: 7 Issue: IV Month of publication: April 2019

DOI: https://doi.org/10.22214/ijraset.2019.4498

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Design and Analysis of Oil Cooling Shell and Tube Type Heat exchanger

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Abstract: Heat exchanger used for heat transfer take place for cooling or heating purpose. Shell and tube type heat exchanger is most commonly used in industrial application. The tube diameter, tube length, shell types etc. are all standardized and are available only in certain sizes and geometry, so the design of a shell-and-tube heat exchanger usually involves a trial and error procedure. A set of Design calculation were carried out to Design Shell And tube type heat exchanger For Screw compressor for cooling the oil and comparison were made between various parameters .By calculating the heat transfer coefficient & Pressure drop by changing parameter we came to know that, which parameter is safe for design of Shell and tube type heat exchanger. In Screw compressor oil used for compression process. Earlier practice in screw compressor oil is cooled by air medium. By using Shell and tube type heat exchanger we can successfully cool the oil by water medium.

This study has been undertaken to study design and analysis of the shell and tube heat exchanger. Shell and tube heat exchangers are found to be a widely used heat exchanger in industry for heat exchange purpose. This study shows the effect of various parameters on shell and tube type heat exchanger such as heat transfercoefficient ,pressure drop , pitch layout and baffle spacing. Standard Design calculations are used to study the same. The study also shows the simulation work carried out using 'Solidworks' for Shell and tube type heat exchanger.

Keywords: shell and tube type heat exchanger, flow simulation, solidworks, heat exchanger design

I. INTRODUCTION

A heat exchanger is a device used to transfer heat between a solid object and a fluid or between two or more fluids. The fluids may be separated by a solid wall to prevent mixing or they may be in direct contact. They are widely used in refrigeration, air conditioning, , chemical plants, petrochemical plants, power stations, petroleum refineries, natural-gas processing, and sewage treatment, in space for transferring heat. The best application of a heat exchanger is found in an internal combustion engine in which a circulating fluid known as engine coolantflows through radiator coils and air flows past the coils, which cools the coolant and heats the incoming air.

A shell and tube heat exchanger is a class of heat exchanger designs. It is the most commonly used heat exchanger in oil refineries and other large chemical processes and also it is suited for higher-pressure applications. As its name suggest, this type of heat exchanger consists of a shell (a large pressure vessel) with a bundle of tubes inside it. One fluid runs through the tubes, and another fluid flows over the tubes (through the shell) to transfer heat between the two fluids.

JK Files Pvt. Ltd was established in 1978. JK files, a subsidiary of Raymond Ltd is today the largest producers of files in the world. They can manufacture according to any specifications provided by the customer. It has an impressive 32% global market share. The products are made from high quality materials and with modern manufacturing processes and conform to international standards.

The compressors used in the manufacturing plant uses an oil for the purpose of lubrication of parts of compressor. Since the temperature of oil gets increased, in earlier practice fan blower is used to lower the temperature of the oil using water. This cooled oil is again used for lubrication of compressor parts.

The heat exchanger is used in different industries for cooling fluid process. This heat exchanger is to be design for cooling of oil which is being circulated within the compressor. For this, the heat exchanger of shell and tube type is designed which wants to make it practicality in industry in cooling the hot oil from compressor.

A lot of research work has been carried out on the heat exchangers and its optimization with respect to thermal performance of such researches is explained in detailed in section II below.



ISSN: 2321-9653; IC Value: 45.98; SJ Impact Factor: 6.887 Volume 7 Issue IV, Apr 2019- Available at www.ijraset.com

II. LITERATURE REVIEW

Andre L.H. Costa and Eduardo M. Queiroz [1] presented a paper which deals with study about the design optimization of shell-andtube heat exchangers. The formulated problem consists of the minimization of the thermal surface area for a certain service, involving discrete decision variables. Additional constraints represent geometrical features and velocity conditions which must be complied in order to reach a more realistic solution for the process task. The optimization algorithm is based on a search along the tube count table where the established constraints and the investigated design candidates are employed to eliminate non optimal alternatives, thus reducing the number of rating runs executed.

Abhishek Arya [2] carried out experimental work on fixed designed STHX and calculate the heat transfer coefficient and effectiveness. Validation is carried out which gives the result comparison with that of experimental result. Here flow parameters are not varied but size and number of tubes are varied and best efficient model is selected as Optimized value

III. METHODOLOGY

It is possible to study the effect of various parameters on the performance of heat exchanger with the advancement in simulation software's. This research work concern with design, analysis of flow simulation of the heat exchanger and studying the effects on various parameters. In order to validate the theoretical calculations flow simulation is carried out in Solidworks software, comparision have been conducted using different mass flow rates of hot and cold fluids on the heat exchanger

A. Details regarding Screw compressor

Specifications of Screw compressor

- 1) Application- Air screw Compressor
- 2) Make- ELGI
- 3) Rated Power- 125HP
- 4) CFM- 570
- 5) Outlet air pressure-7 bar
- 6) Max. air pressure- 8 bar
- 7) Oil type- Crude Lube Oil
- 8) Outlet oil temperature- 80-90°C
- 9) Req. inlet oil temperature- 30 °C
- 10) Oil pressure- 2 bar
- 11) Oil flow rate- 5 LPM

Based on this parameters further design calculations are done using standard analytical reference book.this calculations are mentioned in below chapter 4.

IV. THEORETICAL DESIGN

A. Properties Of Crude Oil (shell side)

Table 4.1 represents the properties of crude oil which is being circulated in shell side of the heat exchanger. it represents the properties of crude oil such as temperature, specific heat, thermal conductivity, density, viscosity.

Table 4.1 Properties of crude off				
Crude Oil	I/P	O/P		
Temperature(°C)	90	30		
Specific Heat(J/KgK)	2110	1988.9		
Thermal conductivity(W/mK)	0.132	0.1355		
Density(Kg/m ³)	787	787.4		
Viscosity(Nm/s ²)	0.0014	0.00487		

Table 4.1	Properties	of crude	oil
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B. Properties Of Water (Tube Side)

Table 4.2 represents the properties of water which is being circulated in tube side of the heat exchanger.it represents the properties of water such as temperature, specific heat, thermal conductivity, density, viscosity.

Table 4.2 Properties of water					
Water	I/P	Avg	O/P		
Temperature(°C)	25	27.6	30.2		
Specific Heat(J/KgK)	4180	4180	4180		
Thermal conductivity (W/mK)	0.6115	0.61826	0.618		
Density(Kg/m ³)	996.4	995.776	995.15		
Viscosity(Nm/s ²)	0.00086	0.00083	0.0008		

Table 4.2 Properties of water

- C. Calculation for Counter Flow
- 1) Heat Transfer Coefficient (q)

 $= m.c_{p}.(T_{in} - T_{out})_{oil}$

=0.075*2052*(90-30)

=9234 W

2) Heat absorbed by water (Water)

 $= m.c_{p}.(t_{in} - t_{out})_{water}$

 $9234 = 0.425*4180*(t_{out} - 25)$

$$\rightarrow t_{out} = 30.2 \circ C$$

3) Logarithmic Mean Temperature Difference (LMTD)

 $\Delta T_{\rm m} = \frac{(T_{\rm in} - t_{\rm out}) - (T_{\rm out} - t_{\rm in})}{\ln(\frac{T_{\rm in} - t_{\rm out}}{T_{\rm out} - t_{\rm in}})}$ $= \frac{(90 - 30.2) - (30 - 25)}{\ln(\frac{90 - 30.2}{2})}$

4) Rand S are considered as dimensionless temperature ratios

 $R = \frac{T_{in} - T_{out}}{t_{out} - t_{in}}$ = $\frac{90 - 30}{30.2 - 25}$ = 11.538 $S = \frac{tout - tin}{T_{in} - t_{in}}$ = $\frac{30.2 - 25}{90 - 25}$ = 0.08

5) The log mean temperature correction factor (Ft) can be given as,

 $F_{4} = \frac{\sqrt{(R^2+1)}ln(\frac{1-S}{1-RS})}{r}$

$$\begin{split} & \left[(R-1) ln \left[\frac{2 - S[R+1 - \sqrt{(R^2 + 1)}]}{2 - S[R+1 + \sqrt{(R^2 + 1)}]} \right] \\ & = \frac{\sqrt{(11.538^2 + 1)} ln (\frac{1 - 0.08}{1 - 11.538 + 0.08})}{(11.538 - 1) ln \left[\frac{2 - 0.08[11.538 + 1 - \sqrt{(11.538^2 + 1)}]}{2 - 0.08[11.538 + 1 + \sqrt{(11.538^2 + 1)}]} \right] \\ & = 0.95 \\ \Delta T_{mc} = F_t * LMTD \\ & = 0.95 * 22.0828 \\ & = 20.97 \circ C \end{split}$$



6) Overall Heat Transfer coefficient Assumed $(U_{assumed}) = 60 \text{ W/m}^2 \circ \text{C}$ 7) Provisional Area q $A = \cdot$ $\overline{U*\Delta T_{mc}}$ 9234 60*20.97 $=7.33m^2 \cong 7.35 m^2$ 8) Tube Dimensions Tube Inner Diameter $(d_i) = 0.012m$ Tube Outer Diameter (d_0)= 0.016m Tube length=1.5m 9) Surface Area $A_{sc} = \prod \times d_o \times L$ = ∏×0.016×1.5 $=0.0754 \text{ m}^2$ 10) No. of Tubes $N = \frac{A}{Asc}$ 7.35 0.0754 = 97.48 ≅ 98 As the Shell side Fluid is relatively clean use Triangular Pitch and considering 1:2 pass (1shell pass and 2 tube passes) (K = 0.249, n = 2.207)11) Bundle Diameter $D_b = d_o \left(\frac{N}{k}\right)^{1/n}$ $= 0.016 \left(\frac{98}{0.249}\right)^{1/2.207}$ =0.2398≅ 0.24m 12) Bundle Diameter Clearance c =0.044.8 m 13) Shell Diameter $D_s = D_b + c$ =0.24 + 0.044.8=0.2848 m D. Tube Fluid film Coefficient Mean Water Temp= $\frac{tout + tin}{c}$ 1) $=\frac{2}{25+30.2}$ 2 =27.6°C 2) Cross Sectional Area Of Tube $A_c = \frac{\prod}{4} \times d_i^2$ $=\frac{\Pi}{4} \times 0.012^2$ $= 0.0001130 \text{ m}^2$ 3) Tube Velocity (Water linear velocity) $(u_w) = \frac{m_w}{N}$ $\rho_W * \frac{N}{2} * Ac$ 0.425 $=\frac{}{995.78 \times \frac{98}{2} \times 0.000113}$ =0.07708 m/s 4) Reynolds Number $\operatorname{Re} = \frac{\rho . u . d_i}{d_i}$ μ



 $995.776 {\times} 0.07701 {\times} 0.012$ 0.00083156 =1106.7 5) Tube Side Heat Transfer Coefficient $h_i = \frac{c_{\rm p.} u_{\rm w}^{0.8} \cdot (1.35 + 0.02 \times \frac{\text{tout + tin}}{2})}{d^{0.2}}$ $d_i^{0.2}$ $4180*0.077^{0.8}*(1.35+0.02\times\frac{30.2+25}{2})$ = 12^{0.2} =622.03 W/m²°C E. Shell Side Coefficient 1) Baffle Spacing $l_B = \frac{Ds}{5}$ $=\frac{0.2448}{5}$ =0.04896 m 2) Tube Pitch $T_p = 1.25 \times d_o$ = 1.25 × 0.016 = 0.02 m3) Cross Flow Area $\mathbf{A}_{\mathrm{s}} = \frac{(T_p - d_o)}{T_p} \times \mathbf{D}_{\mathrm{s}} \times l_B$ $=\frac{(0.02-0.016)}{0.02} \times 0.2848 \times 0.059$ $=0.003244m^2$ 4) Mass Velocity $G_s = \frac{m_o}{A_s}$ 0.075 $=\frac{0.00}{0.003244}$ $=23 \text{ Kg/s m}^2$ 5) Equivalent Diameter $d_e = \frac{1.1}{d_0} (T_p^2 - 0.917 \times d_o^2)$ $= \frac{1.1}{16} \left(20^2 - 0.917 \times 16^2 \right)$ = 11.36 mm = 0.01136 m 6) Reynolds Number $\operatorname{Re} = \frac{G_S * d_e}{d_s * d_e}$ μ =23*0.01136 0.00316 =83.327) Prandtl Number $P_r = \frac{c_p * \mu}{c_p * \mu}$ k_f $=\frac{2052 \times 0.00316}{2052 \times 0.00316}$ 0.134 =48.39Choose 25% baffle Cut 8) $J_{\rm H} = 0.5 \ (1 + \frac{l_B}{D_s})(0.08 \text{Re}^{0.6821} + 0.7 \text{Re}^{0.1772})$ =0.5 $(1 + \frac{1}{5})(0.08*75.7^{0.6821} + 0.7*75.7^{0.1772})$ =1.89



F. Shell Side Heat Transfer Coefficient Can Be Given As

$$I) h_{s} = \frac{k \times J_{H} \times P_{T}^{\frac{1}{3}}}{d_{e}}$$
$$= \frac{0.134 \times 1.88 \times 48.39^{\frac{1}{3}}}{0.01136}$$
$$= 81.53 W/m^{2} \circ C$$

2) Overall Coefficient

$$\frac{1}{U_o} = \frac{1}{h_s} + \frac{1}{h_{od}} + \frac{d_o * ln\left(\frac{d_o}{d_i}\right)}{2k_w} + \left(\frac{d_o}{d_i}\right) * \frac{1}{h_{id}} + \left(\frac{d_o}{d_i}\right) * \frac{1}{h_i}$$
$$\frac{1}{U_o} = \frac{1}{81.57} + \frac{1}{5000} + \frac{0.016 * ln\left(\frac{16}{12}\right)}{2 * 45} + \left(\frac{16}{12}\right) * \frac{1}{3000} + \left(\frac{16}{12}\right) * \frac{1}{622.03}$$
$$U_o = 66.5 \text{W/m}^2 \text{°C}$$
$$(U_o = 66.5) > (U_o \text{ assumed} = 60)$$

G. Pressure Drop 1) Tube side Pressure Drop From Fig (12.24)(R.K.Sinnot) for Re = 1106.7 $J_{\rm f}\,{=}\,0.009$ $\Delta P_t = N_p \times (8 \times J_f(\frac{L}{di}) + 2.5) \times \rho \cdot \frac{\mu t^2}{2}$ $=\!2\!\times\!(8\!\times\!0.009(\tfrac{1.5}{0.012})+2.5)\times995.7*\tfrac{0.077^2}{2}$ $=67.89 \text{ N/m}^2 = 67.89 \text{ Pa}$ 2) Shell Side Pressure Drop For Re = 75.70 from fig 12.30 $J_f = 0.32$ $\Delta P_{s} = 8 \times J_{f} \times (\frac{Ds}{de}) \times (\frac{L}{l_{h}}) \times (\rho \cdot \frac{\mu_{s}^{2}}{2})$ $=8\times J_{f}\times (\frac{Ds}{de})\times (\frac{L}{l_{b}})\times (\rho.\frac{\left\{\frac{G_{s}}{\rho}\right\}^{2}}{2})$ $=8\times0.32\times(\frac{0.284.8}{0.01136})\times(\frac{1.5}{0.059})\times(818.95*\frac{\left\{\frac{21.05}{818.95}\right\}^2}{2})$ $=487.28 \text{ N/m}^2 = 487.28 \text{ Pa}$ H. Component Design 1) Shell Shell Thickness $=\frac{p*D_i}{2f.J-p}+c$.5*284.8 2*130*.75-.5+5 =5.73mm $\cong 10$ mm Shell Outer Diameter = 284.8+20mm =304.8 mm 2) Shell Cover Shell Cover inner diameter Ds=284.8 mm Thickness of torispherical head (h) Ri=284.8 mm r_i=0.06* Ri =0.06* 284.8 =17.0688 mm



$$W = \frac{1}{4} \left(3 + \sqrt{\frac{Ri}{ri}} \right)$$
$$= \frac{1}{4} \left(3 + \sqrt{\frac{284.8}{17.066}} \right)$$
$$= 1.77$$

Corrosion Allowance=3mm

Thickness of a torispherical head

 $t_{h} = \frac{p * Ri * w}{2 * f * j - 0.2 * p} + C$ = $\frac{0.5 * 284.8 * 1.77}{2 * 130 * 0.7 - 0.2 * p} + 3$ = 4.43mm

Inside depth of the head (hi)

$$\begin{aligned} h_{i} = R_{i} - \left[\left(\mathsf{Ri} - \frac{Ds}{2} \right) \left(\mathsf{Ri} + \frac{Ds}{2} \right) + 2ri \right]^{1/2} \\ h_{i} = 284.8 - \left[\left(284.8 - \frac{284.8}{2} \right) \left(284.8 + \frac{284.8}{2} \right) + 2 * 17.688 \right]^{1/2} \end{aligned}$$

=40mm

3) Channel cover Outer diameter=304.8 mm

Thickness (t_{cc})

$$t_{cc} = \frac{Dc}{10} \frac{\sqrt{c1*p}}{f} + c$$
$$= \frac{304.8}{10} \frac{\sqrt{0.3*5}}{13} + 3$$
$$= 5.87 \text{mm}$$

4) Pass partition plate = 10mm

5) Tube sheet thickness $F = Gn \sqrt{n}$

$$tss = \frac{p * 6p}{3} \sqrt{\frac{p}{k*f}}$$
$$= \frac{1.25 * 284..8}{3} \sqrt{\frac{0.5}{0.419 * 130}}$$

6) Nozzle

Nozzle Diameter=50.8 mm (2 inch) Thickness

 $tn = \frac{p * Dn}{dt} + c$

$${}^{2Fj-p} = \frac{0.55*50.8}{2*138*0.85-0.55} + 3$$

=3.11 mm

7) Design of gaskets

Gasket factor (m) =2.5; Maximum design seating stress (Y), kgf/mm²= 2.04

 $\frac{Dog}{Dig} = \sqrt{\frac{Y-pm}{Y-p(m+1)}}$ $\frac{Dog}{285.05} = \sqrt{\frac{2.04-0.056*2.5}{2.04-0.056*3.5}}$ $D_{og} = 290.8mm = 292.8mm$ Width 8mm below
8) Flange Thickness
Flange thickness=25m
9) Bolts
bolt circle diameter (Cb) = 375 mm



ISSN: 2321-9653; IC Value: 45.98; SJ Impact Factor: 6.887 Volume 7 Issue IV, Apr 2019- Available at www.ijraset.com

r=18 mm Root diameter Root Diameter =18 mm 10) Baffles **Baffle Spacing** $l_B = \frac{Ds}{5}$ ______ 5 =0.04896 m Tube vertical pitch $T_p' = 0.87 * T_p$ =87*20 =17.4 mm Baffle cut height Hc=0.25* Ds =0.25*284.8 =71.2mm Height between baffle tips $H_{bt} = D_s - 2 H_c$ =284.8-2*71.2 =142.4mm Number of constrictions crossed <u>H_{bt</u></u>} Ncv= T_p $=\frac{142.4}{17.4}$ =8.183

From Figure 12.32 (R.K.Sinnot) F_n =0.99 for Ncv=8.183 Height from the baffle chord to the top of the tube bundle

 $H_b = \frac{D_b}{2} - D_s (0.5 - B_c)$ $=\frac{240}{2} - 284.8(0.5 - 0.25)$ =48.8mm ≅49 mm Bundle cut = $\frac{H_b}{D_h}$ $=\frac{49}{240}$ =0.2041From Figure 12.41 (R.K.Sinnot) at cut of 0.2041 R'_a =0.14 Tubes in one window area, $N_w = N R'_a$ =98*0.14 =13.72≅14 Tubes in cross-flow area $N_c = N - 2 N_w$ = 98 - 2*14=70Number of baffles Nb= $\frac{L}{-1}$ -1 l_B =<u>1500</u> -1 56.96 =25.33≅25



Table:41 Component			
Specimen	Dimension	Material	
Inner Diameter	284.8 mm	Mild Steel	
Outer Diameter	304.8 mm		
Inner Diameter	16 mm	Stainless Steel A 320	
Outer Diameter	12 mm		
pitch	20 mm		
Baffle Spacing	500 mm	Mild Steel	
No. Of Baffles	2		
Baffle thickness	6 mm		
Thickness	10 mm	Stainless steel	
Crown radius	284.8 mm		
Knuckle radius	20 mm		
Inside Depth	100 mm		
Thickness	10 mm	Stainless steel	
		Soft Al	
Outide diameter	292.8mm		
Thickness	25 mm	Gray Cast Iron HT150	
	1.7		
thickness	15 mm	Gray Cast Iron HT150	
No of Bolts	12	Hot Rolled Carbon Steel	
	12		
Pitch Circle	375 mm		
Root Branneter	10		
	Specimen Inner Diameter Outer Diameter Inner Diameter Outer Diameter pitch Baffle Spacing No. Of Baffles Baffle thickness Thickness Crown radius Knuckle radius	Inner Diameter284.8 mmOuter Diameter304.8 mmInner Diameter16 mmOuter Diameter12 mmpitch20 mmBaffle Spacing500 mmNo. Of Baffles2Baffle thickness6 mmThickness10 mmCrown radius284.8 mmKnuckle radius20 mmInside Depth100 mmThickness10 mmThickness10 mmThickness10 mmThickness20 mmInside diameter284.8 mmOutide diameter292.8mmThickness15 mmNo of Bolts12Pitch Circle375 mm	

Table:41 Components and its material

V. CAD Model





Fig. 5.1 :CAD Design (Tubes and Baffles)



International Journal for Research in Applied Science & Engineering Technology (IJRASET) ISSN: 2321-9653; IC Value: 45.98; SJ Impact Factor: 6.887 Volume 7 Issue IV, Apr 2019- Available at www.ijraset.com



Fig 5.2 :CAD Design (Outer Components)

VI. RESULT

- A. Effect Of Different Parameters On Heat Transfer Coefficient
- 1) Effect of tube outside diameter for same mass flow rate of water.



Sr No.	1	2	3	4	5
Tube	16	20	25	30	38
Outer					
dia(mm)					
Heat Trans	65.29	51	42.13	35.54	27.96
Coeff.					



2) Effect of mass flow rate for same tube outside diameter

Table 6.2 mass flow rate and	l same tube outside diameter
------------------------------	------------------------------

Sr No.	1	2	3	4	5
Mass flow	10	15	20	25	30
rate (LPM)					
Heat	55.9	56.4	60.68	62.72	63.99
Transfer					
Coe.					







Effect of no. of tubes for same mass flow 1 U3)



Table:6.3 No. of tubes for same mass flow rate

Results from flow Simulation В.

Fig 6.4 and Fig 6.5 represents Solidworks flow simulation results for different tube material i.e. steel and copper respectively



Fig.6.4Simulate when, tube material- Steel(K=30w/mk)



ISSN: 2321-9653; IC Value: 45.98; SJ Impact Factor: 6.887 Volume 7 Issue IV, Apr 2019- Available at www.ijraset.com



Fig.6.5 Simulate when, tube material- copper(K=220w/mk

VII. CONCLUSION

From the results of flow simulation of the heat exchanger it can be concluded that the simulation gives results close to those obtained from the theoretical calculations.

From the calculation it can be concluded that the overall heat transfer coefficient gets affected due to various parameters such as baffle spacing and tube pitch layout. It can be found that using triangular pitch overall heat transfer coefficient increases, also it increases with decrease in the baffle spacing.

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